



## **Shear buckling coefficients for longitudinally stiffened-haunched steel plate girders**

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### **Abstract**

In this paper, the critical buckling behavior of haunched steel plate girders subjected to shear loading is investigated in depth. The study is conducted through linear buckling analysis using the finite element method. First, the results are validated with previous results for simply supported rectangular and nonrectangular web panels. Thereafter, an extensive parametric study is conducted to investigate the influence of various geometric parameters including the inclination angle, the panel aspect ratio, the size of the flanges and the presence of longitudinal stiffeners on the buckling coefficients. Finally, the results are employed to develop expressions for the shear buckling coefficients for longitudinally stiffened-haunched steel plate girders taking into account the forementioned parameters.

### **1. Introduction**

In modern construction, sustainability has become an important factor in structural design. In sustainable design, the use of traditional materials is optimized or new materials are employed. In this sense, girder with haunches, also called tapered girders or nonrectangular panels, are increasingly employed particularly in the design of steel bridges curved in elevation. In this case, the girder depth is reduced from intermediate supports (maximum depth) to mid span (minimum depth). Over the last decades, tapered girders have been the focus of several research projects, nevertheless this paper is aimed at analyzing tapered plate girder subject to shear.

Usually, two types of analysis are performed to investigate the shear response of tapered steel girders, ultimate strength analysis and linear buckling analysis. The shear strength of unstiffened tapered girder webs has been investigated experimentally and numerically (Mirambell and Zarate 2000, Zarate and Mirambell 2004, Real et al. 2010, Bedynek et al. 2013, Bedynek 2014, Bedynek et al. 2017, Ibrahim et al. 2020, Sediek et al. 2020). The effect of longitudinal stiffening on the shear strength of tapered plate girders was also investigated (Bedynek et al. 2013, Bedynek 2014). Compared to flat web plates, corrugated plates offer an enhanced out-of-plane stiffness that can increase the shear strength of tapered girders (Hassanein and Kharoob 2014, Hassanein and

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Kharoob 2015, Zevallos et al. 2016). Linear buckling analyses of unstiffened tapered web girder have also been performed (Bedynek et al. 2013, Bedynek et al. 2014, Ibrahim et al. 2020; AbdelAleem et al. 2022). In these linear analyses, the influence of the girder flanges was disregarded.

Recently, the strength of longitudinally stiffened tapered girder web panels subjected to combined bending and shear was investigated experimentally and numerically (Poroustad and Kuhlmann 2018, Poroustad and Kuhlmann 2018, Kuhlmann et al. 2020, Poroustad and Kuhlmann 2021). In these studies, the tapered girders had one horizontal flange and an inclined one, and the longitudinal stiffener was placed either parallel to the upper or lower flange. Furthermore, in the loading protocol the inclined flange was always under compression.

As seen above, there is a still a need to investigate the critical buckling response of longitudinally stiffened-haunched girders subject to shear. Therefore, this paper aims at investigating the linear buckling response of this type of girders. The study is conducted numerically through linear buckling analysis using the finite element method. Firstly, a numerical model is built for simply supported rectangular panel subject to shear loading. Once the model is validated, the geometry is extended to longitudinally stiffened-haunched girders. After that, a parametric analysis is performed to investigate the influence of various geometric variables including the inclination angle, the panel aspect ratio, the size of the flanges and the presence of longitudinal stiffeners on the buckling coefficients. In the end, the results from the parametric study are employed to develop expressions for the shear buckling coefficients for longitudinally stiffened-haunched steel plate girders.

## 2. Numerical modeling

Fig. 1 shows the nomenclature used herein for longitudinally stiffened-haunched plate girders. A finite element model is developed using the software ANSYS (ANSYS 2022). Shell 181 elements, with four nodes and six degrees of freedom on each node are employed to model the girder components (flanges, web and stiffener). A unit load is applied downward in the web at the lower height end, as shown in Fig. 2. Through eigenvalue buckling analysis, critical buckling stresses  $\tau_{cr}$  are computed, and according to the theory of plates stability, buckling coefficients are obtained. The critical buckling shear stress for a simply supported rectangular plate is expressed as

$$\tau_{cr} = k_s \frac{\pi^2 E}{12(1-\nu^2)} \left( \frac{t_w}{h} \right)^2 \quad (1)$$

where  $k_s$  is the shear buckling coefficient,  $E$  is the modulus of elasticity,  $\nu$  is the Poisson's ratio,  $h$  is the web depth, and  $t_w$  is web thickness.

### 2.1 Geometry and parameters

Table 1 presents the variables and their range for the parametric study. This geometry is intended to recreate the behavior of haunched girders near the supports of bridges, in which the inclined flange is under compression and the tension field is developed in the short diagonal. Through the analysis, the longitudinal stiffener is always located from mid height of the lower end ( $h_0/2$ ), and varies its inclination: horizontal, parallel to the inclined flange, and from mid-height ( $h_1/2$ ) to mid-height ( $h_0/2$ ).

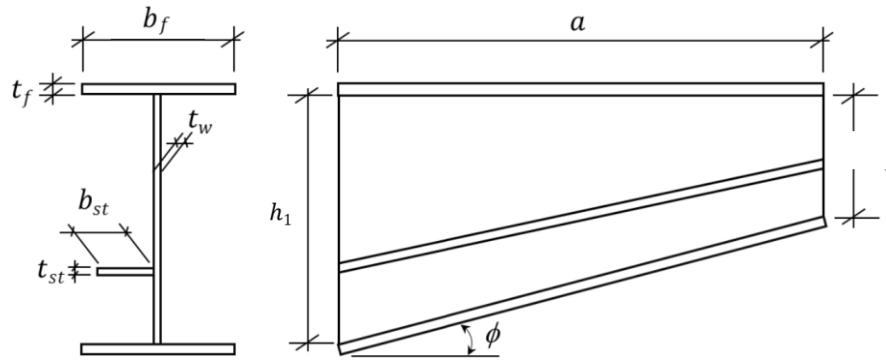


Figure 1: Geometry of longitudinally stiffened-haunched plate girder (Nomenclature)

Table 1: Parameters

Parameters	
Dimension	Value
$h_0$ [mm]	1000
$b_f$	$h_0/6$
$\phi$ [°]	0, 10, 15, 20, 30
$a/h_0$	1, 2, 3
$h_1/t_w$	150, 200, 300
$t_f/t_w$	1, 2, 3
$\gamma_s$	0, 10, 20, 60, 150
Stiffener position	Horizontal, Inclined, Mid-Height
$E$ [MPa]	210000
$\nu$	0.3

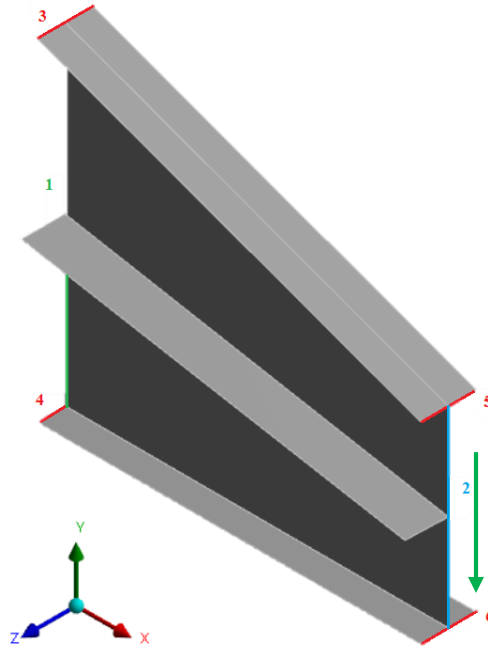


Figure 2: Boundary conditions.

The boundary conditions are shown in Fig. 2 and Table 2. The load is applied downward at the web in the lower end. With this configuration, the tension field is developed in the short diagonal and the inclined flange is under compression.

Table 2: Boundary conditions

Location	Degree of freedom						Force
	U <sub>x</sub>	U <sub>y</sub>	U <sub>z</sub>	R <sub>x</sub>	R <sub>y</sub>	R <sub>z</sub>	
1	R	R	R	R	F	R	0
2	F	F	R	R	F	R	-U <sub>y</sub>
3, 4, 5, 6	F	F	R	R	R	F	0

\*F denotes free and R denotes restrained

## 2.2 Validation procedure

The following material properties were used throughout the study:  $E = 210000$  MPa and  $\nu = 0.3$ . A mesh converge analysis was conducted in Fig. 3 for an unstiffened girder with small flanges, with the following dimensions  $a/h_0 = 1$ ,  $\phi = 15^\circ$ ,  $h_l/t_w = 150$ , and  $t_f/t_w = 1$ . As seen in Fig. 4, the variation in the buckling coefficient is small when for element sizes of 40 mm, hence this element size is chosen for further analysis.

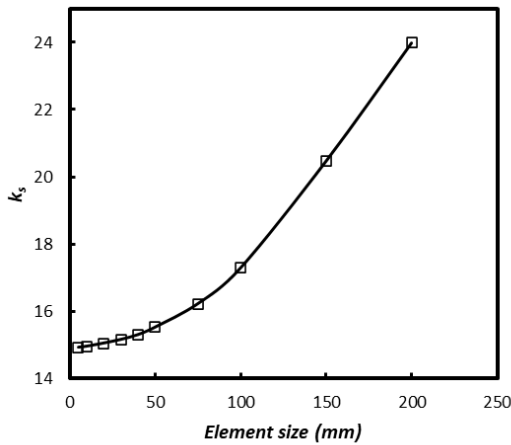


Figure 3. Mesh convergence analysis

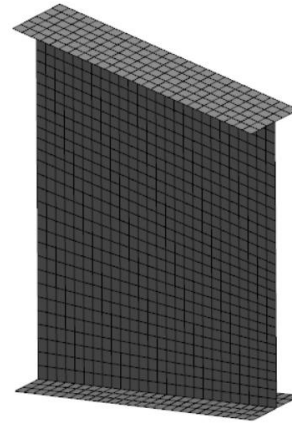


Figure 4. Final mesh

In the literature (Timoshenko and Gere 1936, Ziemian 2010), the shear buckling coefficient for a simply supported rectangular plate subjected to shear is

$$k_s = 5.34 + 4 \left( \frac{h}{a} \right)^2 \quad (2)$$

Also, for non-rectangular simply supported plates, (Bedynek, 2013) proposed four different equations depending on the geometry of the plate and loading direction. For a girder with the inclined flange under compression and the diagonal tension field developed in the short diagonal,

$$k_s = 5.5 \left( \frac{a}{h_w} \right)^{0.8} \tan(\phi) + 8.7 \left( \frac{a}{h_w} \right)^{-0.8} \quad (3)$$

An additional validation is conducted as follows considering a simply supported plate, with dimensions  $a/h_0 = 2$ , and  $h_1/t_w = 200$ . Table 3 presents a comparison between the shear buckling coefficients computed through the finite element model and the those obtained with Eqs. (2) and (3). It is worth pointing out that the value used for  $h$  in Eq. (2) is the major height  $h=h_1$ .

Table 3: Model validation

$\Phi$	$k_s^{FEM}$	$k_s^{Eq.(2)}$	$k_s^{Eq.(3)}$	$k_s^{FEM} / k_s^{Eq.(2)}$	$k_s^{FEM} / k_s^{Eq.(3)}$
0	6.67	6.34	6.59	1.05	1.01
5	8.08	7.17	8.77	1.13	0.92
10	8.78	7.70	9.65	1.14	0.91
20	9.48	8.33	10.46	1.14	0.91
30	10.97	9.98	11.95	1.10	0.92

According to the presented results, with differences between the model and the literature of less than 15%, the model is then validated.

### 3. Parametric study

For the sake of generalization, the effect of five variables in the shear buckling coefficient is studied. The variables are: The flexural rigidity of the stiffener  $\gamma_s$ , the web slenderness  $h_1/t_w$ , the ratio of flange thickness to web thickness  $t_f/t_w$ , the aspect ratio  $a/h_0$  and the haunch inclination  $\tan(\Phi)$ . The different values of the parameters are shown in Table 1, resulting in 2025 cases to run. The analysis of the results is presented next.

#### 3.1 Influence of the stiffener position and flexural rigidity $\gamma_s$

The stiffener flexural rigidity is defined as:

$$\gamma_s = 10.9 \frac{I_{st}}{h_w t_w^3} \quad (4)$$

where  $I_{st}$  is the second moment of area of the stiffener respect to an axis placed at the centroid of the area including the stiffener and a portion of the web of  $15t_w$  on each side as defined by the Eurocode (EC3, 2006). This code also states that when calculating the shear buckling coefficient, the value of the second moment of area must be divided by three.

Fig. 5 shows the variation of the buckling coefficient  $k_s$  in terms of  $\gamma_s$ . The main analysis is that after a reaching a certain value of  $\gamma_s$ , there is not a significant increase of the value of  $k_s$ . This value is the transition between the behavior of a weak stiffener and the behavior of a strong stiffener. As presented in the previous plots, the value of minimum rigidity for a strong stiffener is  $\gamma_s=20$ . After this value the buckling coefficient remain almost constant, therefore the conclusion is that for a strong stiffener, there is no influence of the flexural rigidity on the value of  $k_s$ . Subsequently, the following analyses are performed using only strong stiffeners.

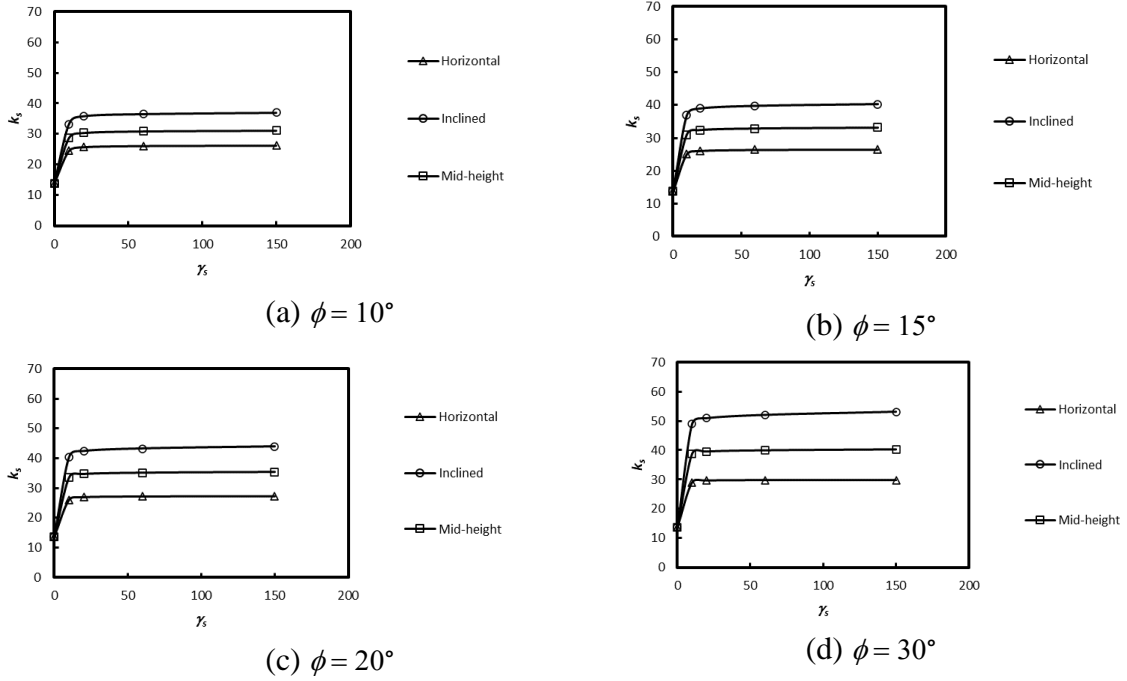


Figure 5: Buckling coefficients  $k_s$  in terms of  $\gamma_s$  ( $a/h_0=1$ ,  $h_1/t_w=300$ ,  $t_f/t_w=1$ ).

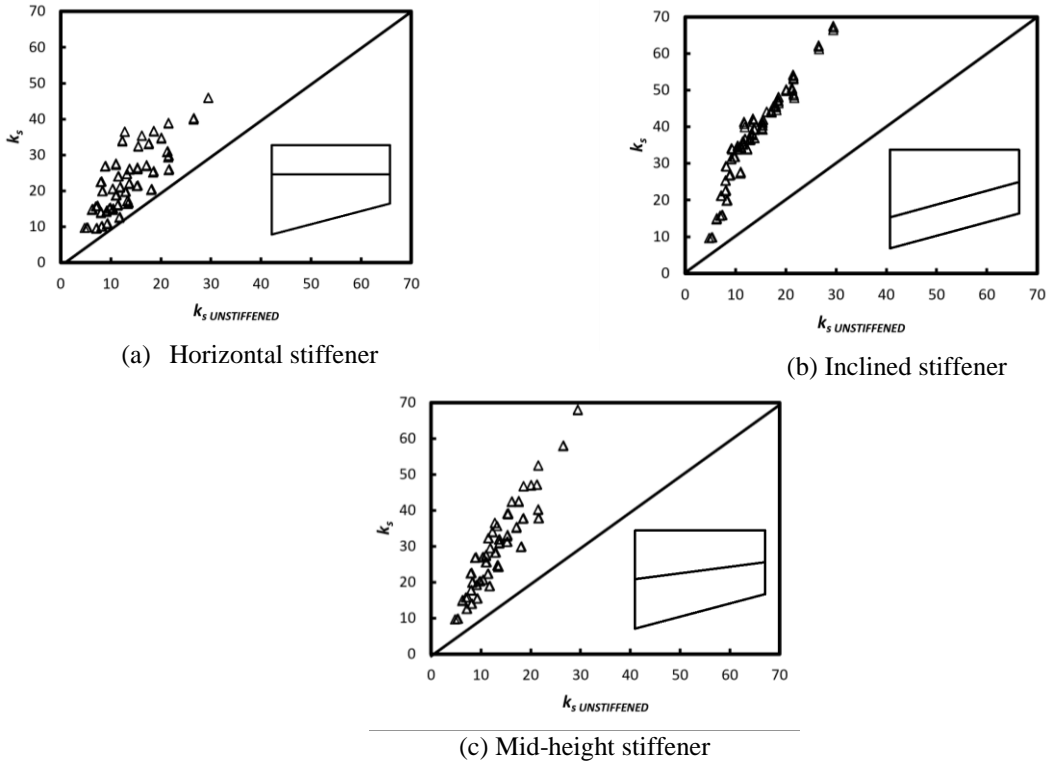


Figure 6: Influence of the stiffener placement.

Fig. 6 shows a significant enhancement in the buckling coefficient with the usage of a longitudinal stiffener. On average, the use of a strong horizontal stiffener increases  $k_s$  by a 78% (Fig. 6a). A

strong stiffener placed parallel to the inclined flange increases  $k_s$  on the average by a 179% (Fig. 6b), and a strong stiffener placed at both of mid heights increases  $k_s$  by a 124% (Fig. 6c).

### 3.2 Influence of the web slenderness $h_1/t_w$

Fig. 7 shows the variation of  $k_s$  with respect to the web slenderness  $h_1/t_w$ , for  $\gamma_s = 150$ ,  $a/h_0 = 2$  and  $t_f/t_w = 2$ .

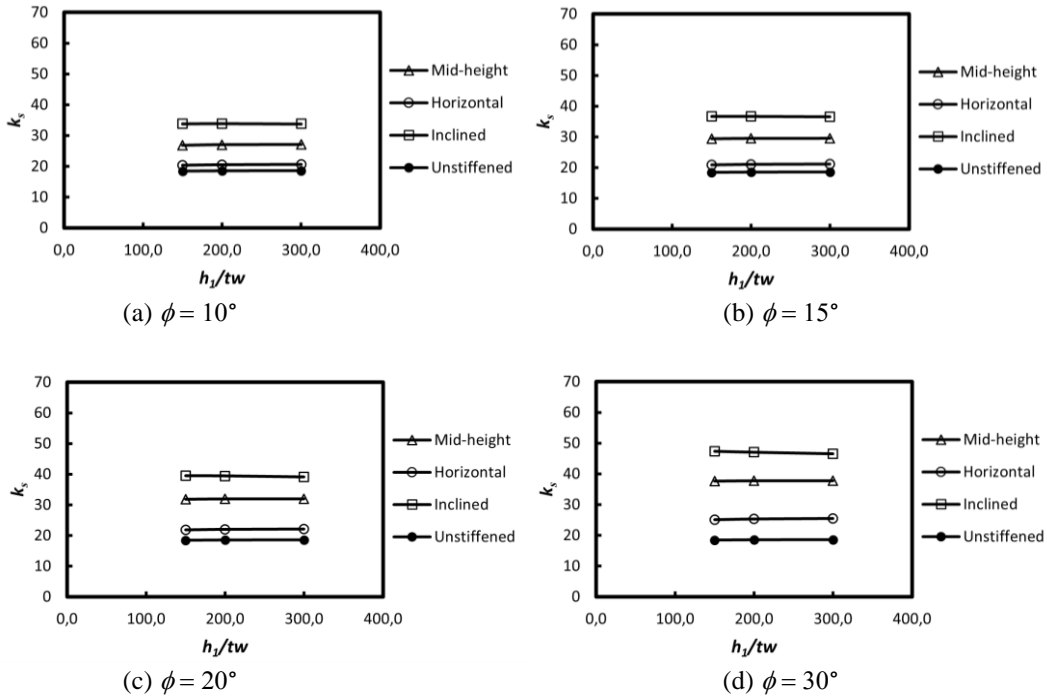


Figure 7: Buckling coefficients  $k_s$  in terms of  $h_1/t_w$  ( $\gamma_s = 150$ ,  $a/h_0 = 2$ ,  $t_f/t_w = 2$ )

It is clearly observed that the slenderness parameter  $h_1/t_w$  has a diminished impact on the shear buckling coefficient, as long as the web is slender  $h_1/t_w > 150$  and the code limit is maintained  $h_1/t_w < 300$  (AASHTO 2020). For this reason, the following results will not include the aforementioned parameter.

### 3.3 Influence of the thickness ratio $t_f/t_w$

From Fig. 8 to Fig. 11 the variation of buckling coefficient  $k_s$  in terms of  $t_f/t_w$  is shown.

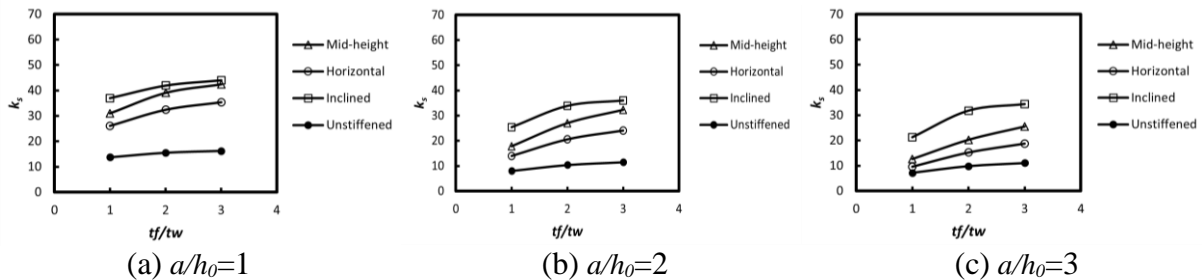


Figure 8: Buckling coefficients  $k_s$  in terms of  $t_f/t_w$  ( $\gamma_s = 150$ ,  $\phi = 10^\circ$ )

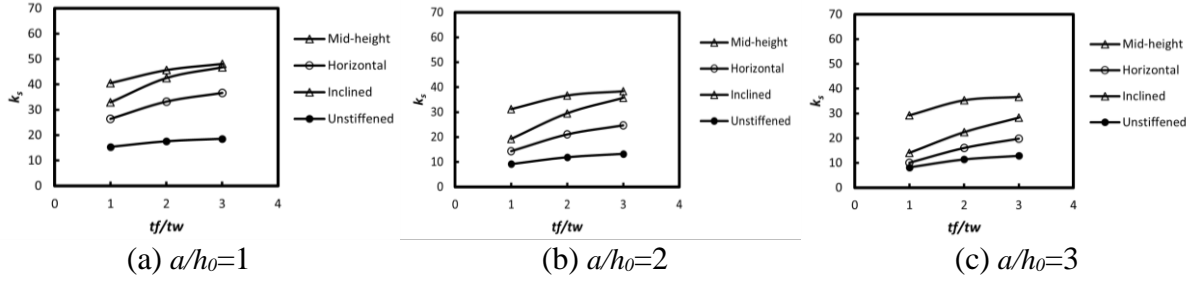


Figure 9: Buckling coefficients  $k_s$  in terms of  $t_f/t_w$  ( $\gamma_s = 150$ ,  $\phi = 15^\circ$ )

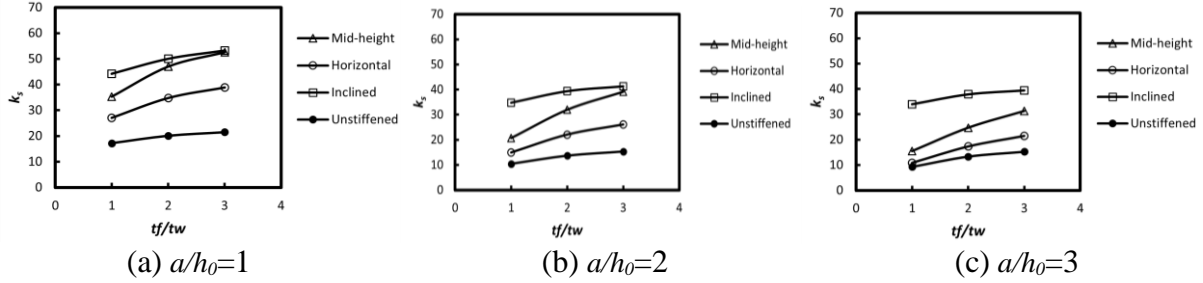


Figure 10: Buckling coefficients  $k_s$  in terms of  $t_f/t_w$  ( $\gamma_s = 150$ ,  $\phi = 20^\circ$ )

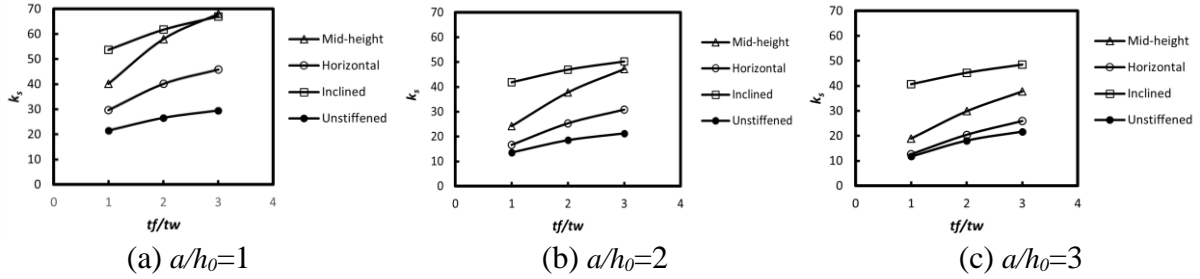


Figure 11: Buckling coefficients  $k_s$  in terms of  $t_f/t_w$  ( $\gamma_s = 150$ ,  $\phi = 30^\circ$ )

It is observed that for all cases, there is an increase in the buckling coefficient as the flange-to-web thickness ratio is greater. This is expected, as the thickness of the flanges increase, the boundary condition of the web to flange juncture approaches to a fixed support, thus increasing the rigidity and the shear capacity of the girder. As the variation of this ratio makes the buckling coefficient change, this value is included in the analysis.

### 3.4 Influence of the panel aspect ratio $a/h_0$

From Fig. 12 to Fig. 15, the variation of the buckling coefficient  $k_s$  in terms of the aspect ratio  $a/h_0$  is presented.

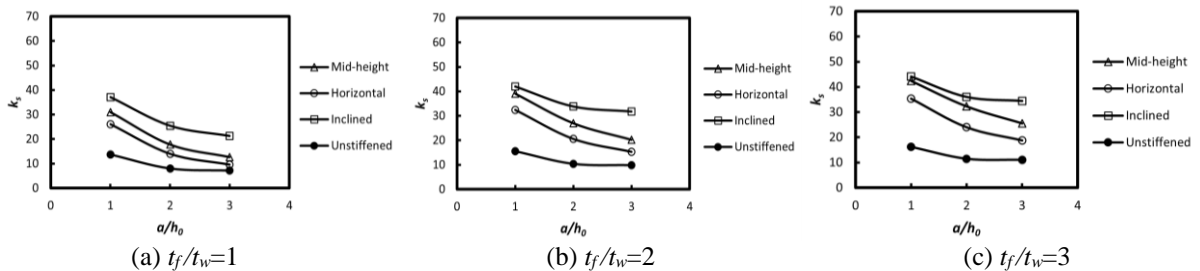


Figure 12: Buckling coefficients  $k_s$  in terms of  $a/h_0$  ( $\gamma_s = 150$ ,  $\phi = 10^\circ$ )



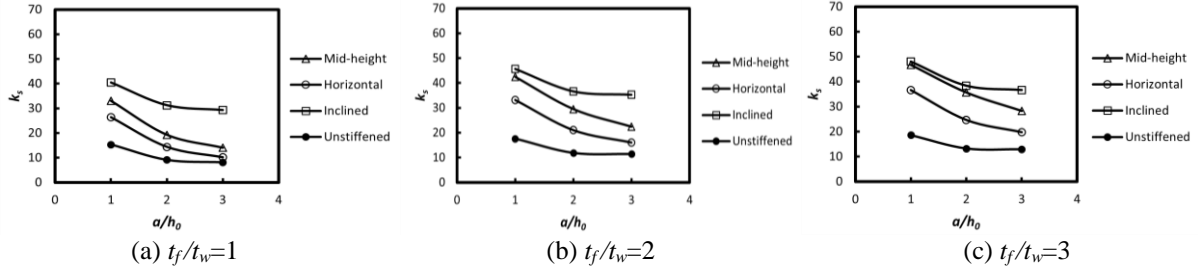


Figure 13: Buckling coefficients  $k_s$  in terms of  $a/h_0$  ( $\gamma_s=150$ ,  $\phi=15^\circ$ )

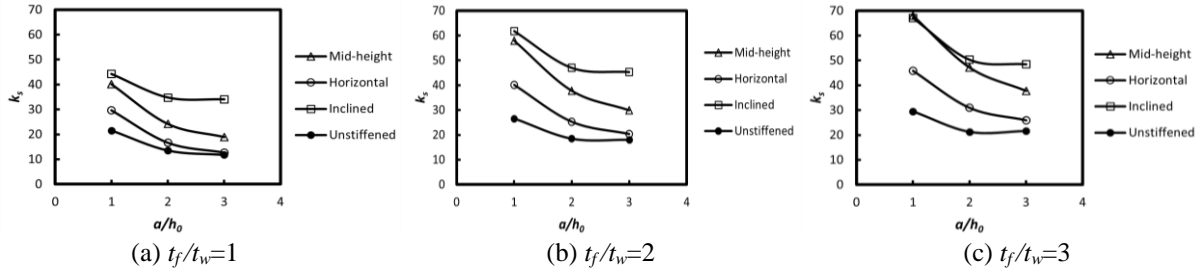


Figure 14: Buckling coefficients  $k_s$  in terms of  $a/h_0$  ( $\gamma_s=150$ ,  $\phi=20^\circ$ )

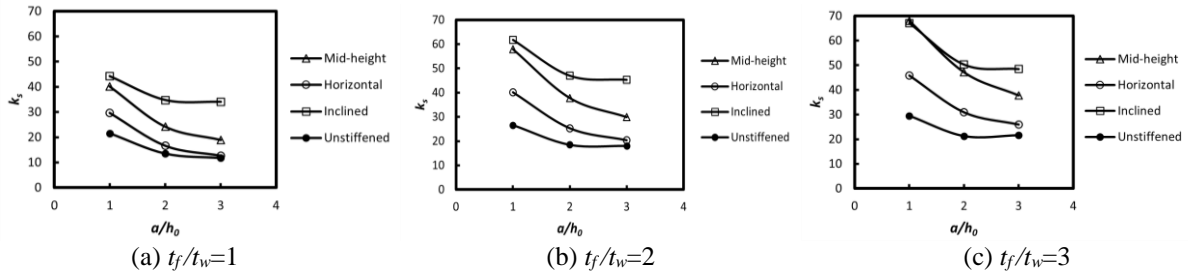


Figure 15: Buckling coefficients  $k_s$  in terms of  $a/h_0$  ( $\gamma_s=150$ ,  $\phi=30^\circ$ )

It is observed that for all cases, as the aspect ratio increases, the buckling coefficient decreases. This is explained because as the panel aspect ratio increases, the panel is longer and bending stresses increase, thus reducing shear buckling capacity. As this parameter makes the buckling coefficient change, this value must be included in the analysis.

### 3.5 Influence of the haunch inclination $\tan(\phi)$

From Fig. 16 to Fig. 18 the variation of the buckling coefficient in terms of the haunch inclination  $\tan(\phi)$  is presented.

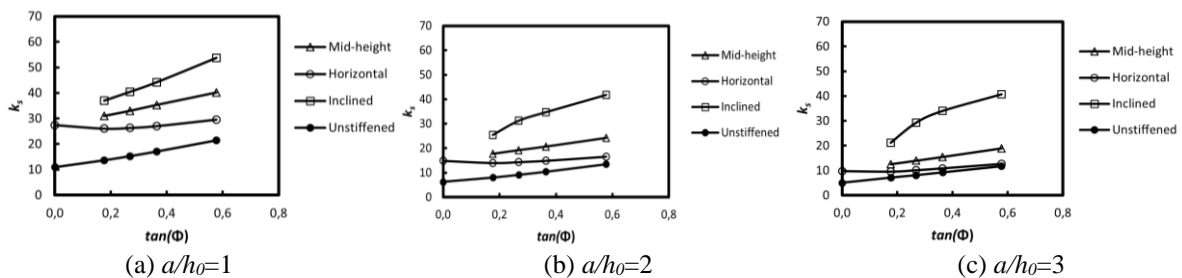


Figure 16: Buckling coefficients  $k_s$  in terms of  $\tan(\phi)$  ( $\gamma_s=150$ ,  $t_f/t_w=1$ )

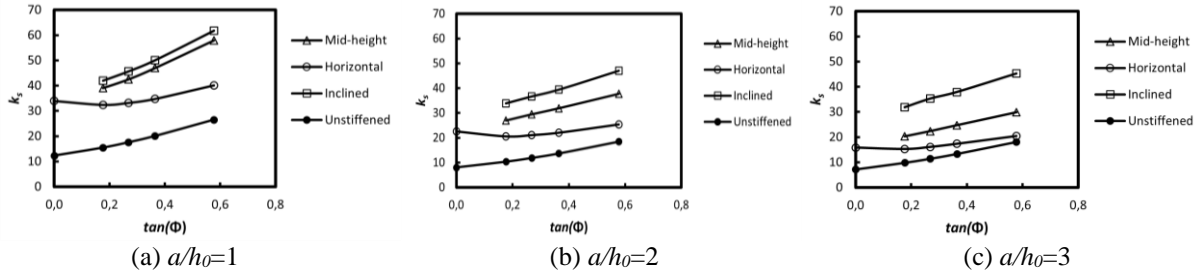


Figure 17: Buckling coefficients  $k_s$  in terms of  $\tan(\phi)$  ( $\gamma_s=150$ ,  $t_f/t_w=2$ )

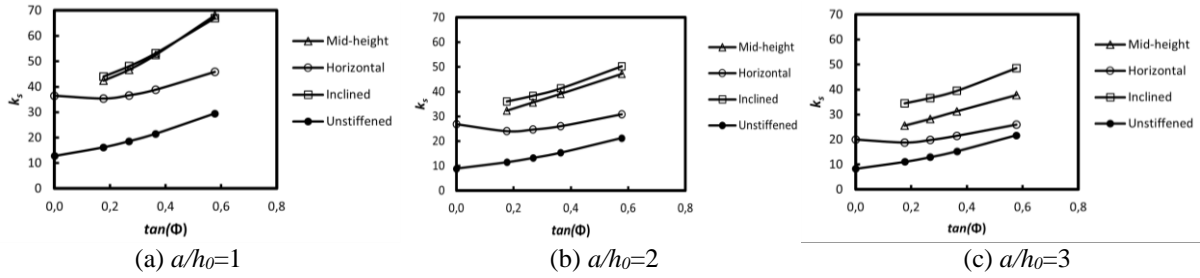


Figure 18: Buckling coefficients  $k_s$  in terms of  $\tan(\phi)$  ( $\gamma_s=150$ ,  $t_f/t_w=3$ )

It is observed that for all cases, as the haunch angle  $\phi$  increases the buckling coefficient also increases. As this parameter affects the buckling coefficient, this value must be included in the analysis.

### 3.5 Buckling shapes.

Fig. 19 shows the buckling shapes, varying the stiffener rigidity and position, for a girder with  $a/h_0=2$ ,  $t_f/t_w=2$ ,  $h_1/t_w=200$ ,  $\phi=20^\circ$ . Observing the deformed shapes in Fig. 19, the following findings are highlighted:

1. When the panel is unstiffened  $\gamma_s=0$ , the buckling shape involves the whole web panel, thus reducing the capacity ( $k_s = 13.70$ ).
2. When the stiffener is weak  $\gamma_s = 10$ , the stiffener does not provide a nodal line of near zero out-of-plane displacement, and the buckling also involves the whole web panel and the stiffener. Nevertheless, the buckling coefficient  $k_s$  increases compared to the unstiffened girder.
3. When the stiffener is strong,  $\gamma_s = 20$ , buckling occurs only in the larger sub-panel, as the stiffener restricts the out-of-plane displacement similar to a nodal line with near zero displacements, thus dividing the whole webpanel into two sub-panels, and increasing the capacity of the girder to resisting shear buckling. For  $\gamma_s = 150$  the difference in the buckling coefficient is very small compared to that obtained for a stiffener with  $\gamma_s = 20$

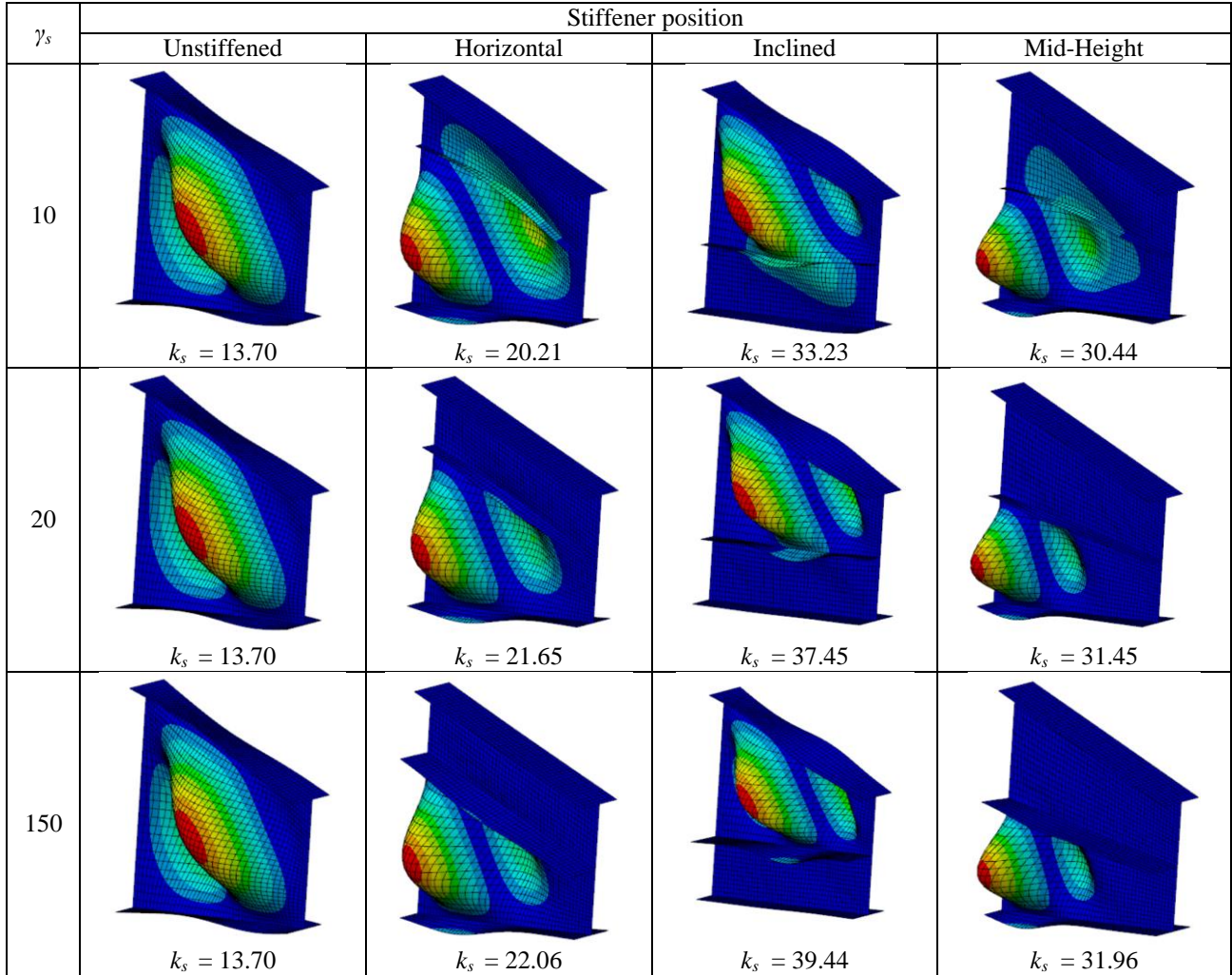


Figure 19: Buckling shapes for  $a/h_0=2$ ,  $t_f/t_w=2$ ,  $h_l/t_w=200$ ,  $\phi = 20^\circ$ .

#### 4. Proposal of buckling coefficient

##### 4.1 Procedure

The following procedure was used to develop the prediction models for each of the stiffener positions: horizontal, inclined and Mid-Height.

1. Determine the variables that may influence in the estimation of the buckling coefficient. The selected variables were:  $t_f/t_w$ ,  $a/h_0$  and  $\tan(\phi)$ . It was found that the influence of the web slenderness ratio,  $h_l/t_w$ , can be neglected.
2. It was established that all the equations were going to be set for strong stiffeners.
3. Construct the dependence charts, for all the possible combinations of angles and  $t_f/t_w$ , it revealed the relationships between the aspect ratio  $a/h_0$  and the buckling coefficient.
4. Once the trendlines of the charts were established, in the form  $y(x)=C_o(x)^\alpha$ , where  $y(x)$  corresponds to the buckling coefficient and  $(x)$  to the aspect ratio, the coefficients were tabulated ( $C_o$  vs  $\alpha$ ).
5. The relationship between  $C_o$  and  $\alpha$  with the parameter  $\tan(\phi)$  were found plotting them again as shown from Fig. 20 to Fig. 22.

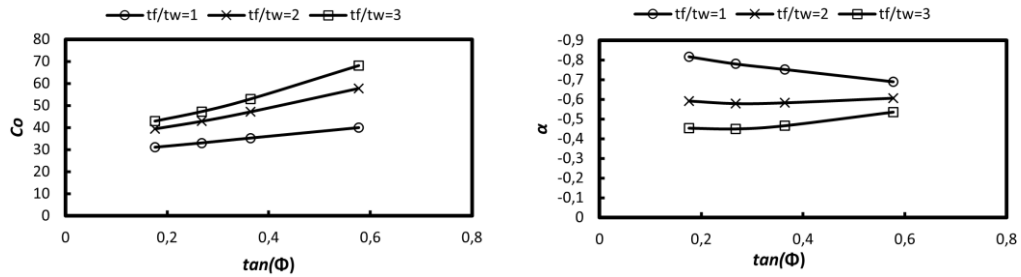


Figure 20:  $C_o$  and  $\alpha$  in terms of  $\tan(\phi)$  for Mid-Height position.

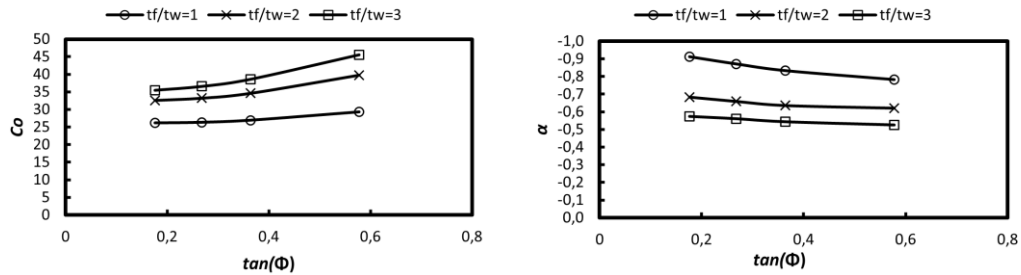


Figure 21:  $C_o$  and  $\alpha$  in terms of  $\tan(\phi)$  for horizontal position.

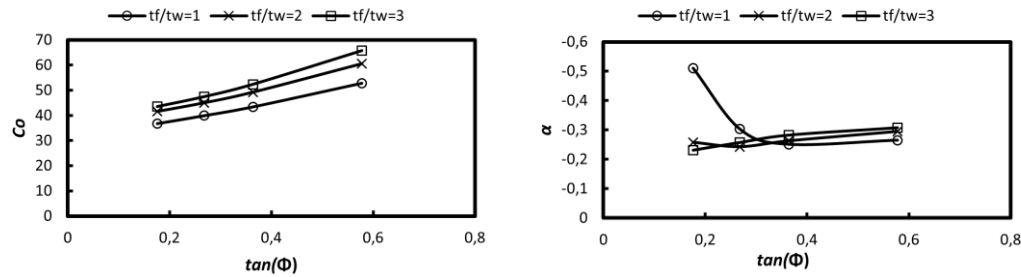


Figure 22:  $C_o$  and  $\alpha$  in terms of  $\tan(\phi)$  for inclined position.

6. Polynomial trendlines were made again to fit the formulas that describe  $C_o$  and  $\alpha$ .
7. A new set of variables  $a$  and  $b$  were plotted in terms of thickness ratio  $t_f/t_w$ , in order to involve all the parameters affecting the buckling coefficient. The plots are shown from Fig. 23 to Fig. 25.

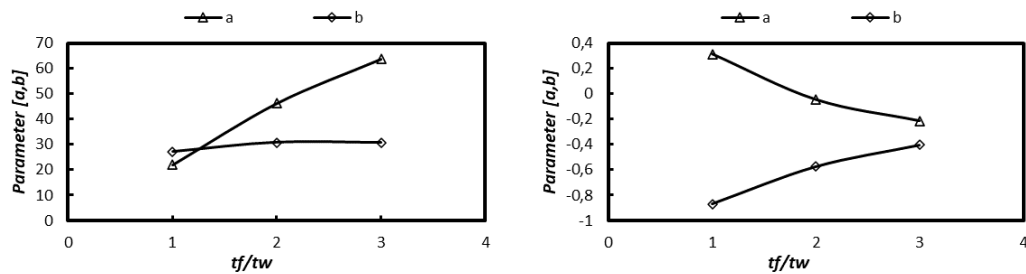


Figure 23:  $a$  and  $b$  in terms of  $t_f/t_w$  for mid-height position.

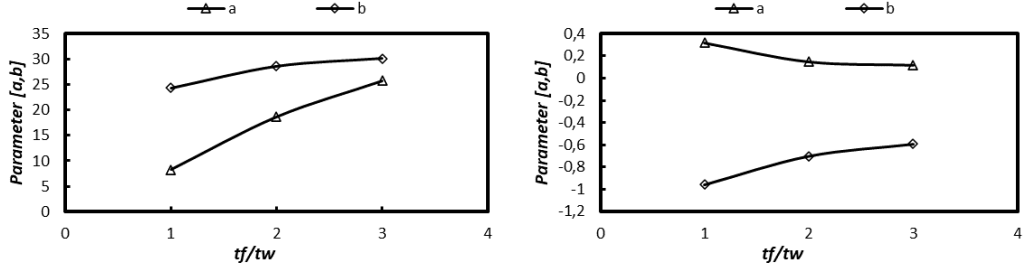


Figure 24:  $a$  and  $b$  in terms of  $t_f/t_w$  for horizontal position.

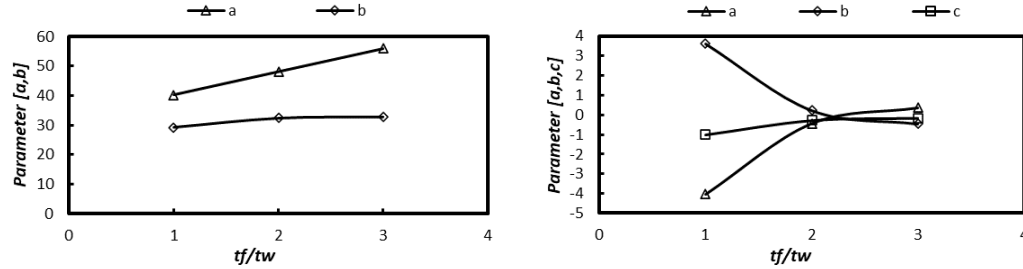


Figure 25:  $a$  and  $b$  in terms of  $t_f/t_w$  for inclined position.

8. In a hierarchical way, the equation proposed in 3) was rewritten with steps 4) and 5), in order to obtain the final equations.

Mid-Height:

$$k_s = \left[ \left( 20.8 \cdot \frac{t_f}{t_w} + 2.4 \right) \cdot \tan\theta + 1.8 \cdot \frac{t_f}{t_w} + 26 \right] \cdot \left( \frac{a}{h_0} \right)^{\left[ \left( -0.2 \cdot \frac{t_f}{t_w} + 0.5 \right) \cdot \tan\theta + 0.2 \cdot \frac{t_f}{t_w} - 1.1 \right]} \quad (5)$$

Horizontal:

$$k_s = 0.95 \cdot \left[ \left( 8.7 \cdot \frac{t_f}{t_w} + 0.1 \right) \cdot \tan\theta + 2.9 \cdot \frac{t_f}{t_w} + 22 \right] \cdot \left( \frac{a}{h_0} \right)^{\left[ \left( -0.1 \cdot \frac{t_f}{t_w} + 0.4 \right) \cdot \tan\theta + 0.2 \cdot \frac{t_f}{t_w} - 1.1 \right]} \quad (6)$$

Inclined:

$$k_s = \left[ \left( 7.9 \cdot \frac{t_f}{t_w} + 32.4 \right) \cdot \tan\theta + 1.8 \cdot \frac{t_f}{t_w} + 28 \right] \cdot \left( \frac{a}{h_0} \right)^{(-0.29)} \quad (7)$$

#### 4.2 Statistical evaluation

Fig. 26 shows the correlation between the computed buckling coefficient and those obtained with Eqs. (5), (6) and (7). It can be seen that the best correlation is attained for the girder stiffened at mid-height with  $R^2=0.99$ , for the stiffener placed horizontally  $R^2=0.95$ , and finally for the stiffener placed parallel to the inclined flange  $R^2=0.91$ .

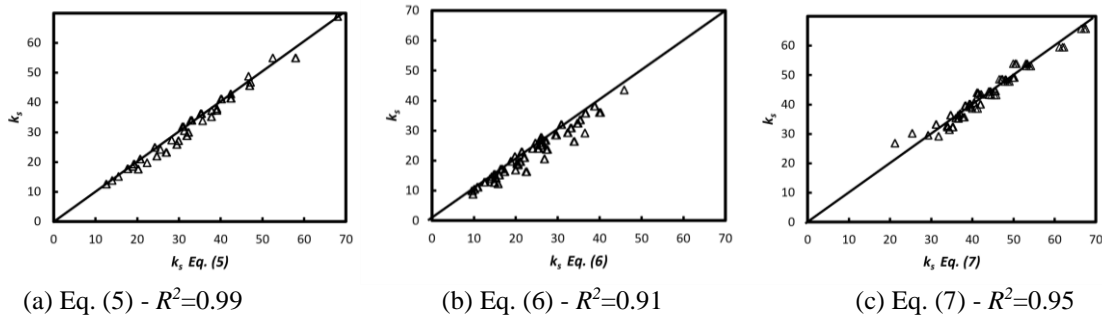


Figure 26. Correlation between numerical results and proposed formulas

## 5. Conclusions

In this paper, the critical buckling response of longitudinally stiffened-haunched steel plate girders subjected to shear loading was investigated through finite elements analysis. An extensive parametric study was performed varying the panel aspect ratio, the flange-to-thickness ratio, the stiffener rigidity and location (mid height, horizontal and inclined). From the results the following conclusion are drawn:

- .- The use of longitudinal stiffeners increases the critical buckling coefficient  $k_s$ , and hence the critical stresses, of haunched girders subjected to shear loading.
- .- Among the investigated parameters, the slenderness of the panel  $h_1/t_w$  has a negligible impact on the buckling coefficients.
- .- Regarding the flexural rigidity of the stiffener, within the range of parameters evaluated herein there is a diminished influence for  $\gamma_s \geq 20$ . Beyond this limit, buckling of the webpanel is divided into two subpanels, and the whole girder attains a greater buckling load. The highest buckling coefficients were attained when the longitudinal stiffener was located parallel to the inclined flange.
- .- Locating the stiffener parallel to the inclined flange works the best.
- .- Buckling coefficients calculated with the proposed formulas attained very high correlation with the values computed numerically.

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